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# TROPOSPHERIC NO, MAPPING USING DOAS FROM AN UAV

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# ABSTRACT

We present the concept of a system for NO<sub>2</sub> mapping at high spatial resolution using a compact UV-VIS spectrometer onboard an Unmanned Aerial Vehicle (UAV). Boundary layer NO<sub>2</sub> columns will be retrieved using the DOAS technique. The aim of this project is to obtain NO<sub>2</sub> maps with a pixel size of a few hundred meters using the whiskbroom scanning geometry. This instrument will be a useful tool for data validation from satellite nadir sensors such as OMI or GOME-2 as well as for assessment of air quality models. Preliminary aspects including flight details (altitude, velocity) and signal-to-noise ratio are also considered. The first experiment is scheduled for the summer of 2012 in Romania.



About NO2



The Nitrogen Dioxide (NO<sub>2</sub>) molecule consists of two oxygen atoms each attached to a nitrogen atom by covalent bonds, having a bond length of 0.120nm.

Fig.1. Skeletal formula of NO<sub>2</sub>

NO<sub>2</sub> plays an important role in the creation and destruction of Ozone  $(O_3)$ (see below), and is known as  $O_3$ precursor. It is also readily photolised into NO and an Oxygen atom.

- The forward and backward reaction of <u>NO<sub>2</sub> natural equilibrium</u>
- $NO_2 + h\nu \rightarrow NO + O(\lambda < 424 nm)$
- $O+O_2 \rightarrow O_3$
- $NO+O_3 \rightarrow NO2+O_2$

**NO<sub>2</sub> production from CO oxidation :** 

 $CO + OH \rightarrow CO_2 + H$  $H+O_2 + M \rightarrow HO_2 + M$  $HO_2 + NO \rightarrow OH + NO_2$  $NO_2 + h\nu \rightarrow NO + O$  $O + O_2 + M \rightarrow O_3 + M$ 



Fig. 2. Block diagram of natural cycle of NO2

#### Natural sources of NO<sub>2</sub>:

- Lightning
- Microbial activity in soil

Anthropogenic sources of NO<sub>2</sub>:

- Transports
- Biomass burning
- Power generation
- Industrial processes

## **UAV DESCRIPTION**

**UAV Specifications:** 

-wing aperture :1,2-2,4m (depending on the operational requirements of the project); -altitude:4,000 meters max, min 2-3 feet;

## **About DOAS**

**Instrumental effects** 

**Beer-Lambert Law in the atmosphere:** 



**Rayleigh and Mie scattering Several absorbers** 

Idea of DOAS: separation of broad and narrow band absorption enables quantitative analysis for some gases, e.g. NO<sub>2</sub>.

The result of the DOAS fit is a Differential Slant Column Density (DSCD) of  $NO_2$  which is the difference between the NO<sub>2</sub> SCDs of the measured spectrum and the Fraunhofer reference spectrum.

The spectrum shown in Fig. 3 was recorded from an ultra light aircraft (ULM) in limb geometry (looking to the horizon) above the Po Valley in Italy, which always shows high level of NO<sub>2</sub>.

Software used for DOAS retrieval: QDoas.

Fig. 3. Example of a NO<sub>2</sub> DOAS retrieval from A. Merlaud et al. paper in preparation for Atmospheric Measurements Technique, 2012



## **SCHEMATIC SET-UP & TECHNICAL DESCRIPTION OF PAYLOAD**

-cruising speed:80-100km/h; -maximum speed:140km/h. Minimum speed:8-10km/h;

-range:10-15km package standard package extended 50km (radius may be increased in accord with the operational requirements of the project); -autonomy:60 minutes.







Fig.5. Image of an UAV



It=Tsc/Np





FOV= $2^{H^{tan}(\beta/2)}$ Tsc – Time of a complete scan Tsc=Ps/V Ps – Pixel size V – Speed Np=FOV/Ps

**OMI NO2 MAP** 

2 3 4 6 8 11 15 20

Steel factory

Weight: 716 g (with no cover ~400g) Spectral range :200-750nm Resolution (FWHM): 1.5 nm Focal length: 75mm Entry slit: 50µm Grating: 600l/mm

NO<sub>2</sub> trop. column [10<sup>15</sup> molec./cm<sup>2</sup>]

Fig. 7. Some components of the payload

#### **Essential requirements for payload and flight experiment**

Total weight: <900g Power: <20W Flight altitude: >2,000m Volume: 300x75x110 mm Flight duration: 60 minutes Data processing: after flight.



Fig. 8. The whisk broom scanning\* \*From Sabins, Jr., F.F., Remote Sensing: Principles and Interpretation, 2nd Ed., W.H. Freeman

#### Table 1.Relations between parameters used

| Integration<br>time<br>[sec] | S/N  | Altitude<br>[m] | Swath<br>[km] | Pixel's<br>size<br>[m] | Speed<br>[km/h] | Complete<br>scan<br>[sec] | Number of<br>pixels along the<br>track |
|------------------------------|------|-----------------|---------------|------------------------|-----------------|---------------------------|--|
| 6                            | ~10  | 3000            | 10            | 1000                   | 60              | 60                        | 10                                     |
| 4.28                         | 8    | 4000            | 14            | 1000                   | 60              | 60                        | 14                                     |
| 1.5                          | ~5   | 3000            | 10            | 500                    | 60              | 30                        | 20                                     |
| 1.07                         | 4    | 4000            | 14            | 500                    | 60              | 30                        | 28                                     |
| 0.75                         | ~3.5 | 3000            | 10            | 250                    | 30              | 30                        | 40                                     |
| 0.26                         | ~2   | 4000            | 14            | 250                    | 60              | 15                        | 56                                     |
| 0.12                         | ~1.5 | 3000            | 10            | 100                    | 30              | 12                        | 100                                    |
| 0.085                        | ~1   | 4000            | 14            | 100                    | 30              | 12                        | 140                                    |

It – Integration time Np – Number of pixels

Where: H – Altitude of UAV FOV – Field of View

 $\beta$  – scanning angle (120°)

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Fig. 10. The NO<sub>2</sub> map over Romania. OMI shows an important smoothing effect due to the size of the pixel

**Calculation of the Volume Mixing Ratio** in the Boundary Layer from a Slant Column

Brăila

Date:01.SEPT.2011 Time:10'00(GMT+2) SZA=45°; BLH=1000m SC=2.52e16molec/cm<sup>2</sup>  $\rho_{air}$ = 2.42e19molec/cm<sup>3</sup>

**AMF=2.41** VC=1.04e16molec/cm<sup>2</sup> C=1.04e11molec/cm<sup>3</sup> VMR=4.29e-9molec/cm<sup>3</sup> =4.29ppb

VC=SC/AMF; VMR= $C_{NO2}/\rho_{air}$ C=VC/BLH AMF=1+1/cos(SZA) Where: SC – Slant Column VC – Vertical Column AMF – Air Mass Factor C – Concentration BLH – Boundary Layer Height VMR – Volume Mixing Ratio

SZA – Solar Zenith Angle

Fig. 11. The NO<sub>2</sub> plot from glider & DOAS experiment



Fig. 12. Image from glider & DOAS experiment

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